

SUPPLEMENTAL TYPE CERTIFICATE

10065657

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

ELECTRONICS INTERNATIONAL, Inc.

63296 POWELL BUTTE HIGHWAY
BEND OR 97701
USA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: SEE FAA APPROVED MODEL LIST
(AML) SA02402SE,
dated 24 June 2014.

Type Certificate Holder: See FAA AML
Type: See FAA AML

Model: See FAA AML

Original STC Number: FAA STC SA02402SE

Description of Design Change:

Installation of Electronics International Inc. Glass Panel Engine Monitor (with In-Flight Data) CGR-30C.

EASA Certification Basis:

The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.


Associated Technical Documentation:

1) Master drawing list as specified on AML SA02402SE, or later FAA approved revision.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Cologne, Germany, 29 May 2018



Dominique ROLAND
Head of General Aviation and
Remotely Piloted Aircraft Systems (RPAS)



2) Electronics International, Inc. Instructions for Continued Airworthiness, Document No. 06211301, Revision E, dated 28 March 2014, or later FAA accepted revision.

Limitations/Conditions:

1) Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

2) The STC is valid for aircraft types and models approved by EASA and listed on the FAA approved AML SA02402SE, dated 24 June 2014.

- End -

